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APPLICATION NO.	FILING DATE	FIRST NAMED I	FIRST NAMED INVENTOR		ITORNEY DOCKET NO.
09/328,931	06/09/99	MORRIS		D	TES-7356-003
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TIMOTHY E SIEGEL PATENT ATTORNEY		200,000		VERDIER, C	
	CADAM AVENL	IE		ART UNIT	PAPER NUMBER
SUITE 200				3745	15/
PORTLAND O	R 97201			DATE MAILED:	06/05/01

Please find below and/or attached an Office communication concerning this application or proceeding.

Commissioner of Patents and Trademarks

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Office Action Summary		Application No.	Applicant(s)					
		09/328,931	MORRIS, DAVID CURT					
		Examiner	Art Unit					
		Christopher Verdier	3745					
	The MAILING DATE of this communication appears on the cover sheet with the correspondence address Period for Reply							
THE N - Exten after 3 - If the - If NO - Failur - Any re	ORTENED STATUTORY PERIOD FOR REPLY MAILING DATE OF THIS COMMUNICATION. sions of time may be available under the provisions of 37 CFR 1.13 SIX (6) MONTHS from the mailing date of this communication. period for reply specified above is less than thirty (30) days, a reply period for reply is specified above, the maximum statutory period we to reply within the set or extended period for reply will, by statute, eply received by the Office later than three months after the mailing d patent term adjustment. See 37 CFR 1.704(b).	36 (a). In no event, however, may a reply be to within the statutory minimum of thirty (30) day fill apply and will expire SIX (6) MONTHS from cause the application to become ABANDONE	mely filed s will be considered timely. the mailing date of this communication. D (35 U.S.C. § 133).					
1)⊠	Responsive to communication(s) filed on 23 M	<u> March 2001</u> .						
2a)⊠	This action is FINAL . 2b) ☐ Thi	is action is non-final.						
3)	Since this application is in condition for allowance except for formal matters, prosecution as to the merits is closed in accordance with the practice under <i>Ex parte Quayle</i> , 1935 C.D. 11, 453 O.G. 213.							
Dispositi	on of Claims							
4)⊠	Claim(s) <u>1-4</u> is/are pending in the application.							
4a) Of the above claim(s) is/are withdrawn from consideration.								
5) 🗌								
6)⊠	6)⊠ Claim(s) <u>1-4</u> is/are rejected.							
7)	7) Claim(s) is/are objected to.							
8)[Claims are subject to restriction and/or	election requirement.						
Applicati	on Papers							
9)⊠ The specification is objected to by the Examiner.								
10)⊠	10)⊠ The drawing(s) filed on <u>6-9-99</u> is/are objected to by the Examiner.							
11)☐ The proposed drawing correction filed on is: a)☐ approved b)☐ disapproved.								
12) The oath or declaration is objected to by the Examiner.								
Priority u	ınder 35 U.S.C. ≬ 119							
13) Acknowledgment is made of a claim for foreign priority under 35 U.S.C. § 119(a)-(d) or (f).								
a) All b) Some * c) None of:								
1. Certified copies of the priority documents have been received.								
	2. Certified copies of the priority documents have been received in Application No							
 3. Copies of the certified copies of the priority documents have been received in this National Stage application from the International Bureau (PCT Rule 17.2(a)). * See the attached detailed Office action for a list of the certified copies not received. 								
14) Acknowledgement is made of a claim for domestic priority under 35 U.S.C. § 119(e).								
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Attachment	t(s)							
15)	ce of References Cited (PTO-892) ce of Draftsperson's Patent Drawing Review (PTO-948) rmation Disclosure Statement(s) (PTO-1449) Paper No(s)	19) Notice of Informa	ary (PTO-413) Paper No(s) Il Patent Application (PTO-152)					

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Applicant's Amendment dated March 23, 2001 has been carefully considered but is deemed non-persuasive. Claims 1-4 are pending, with claims 3-4 being newly added.

Applicant has argued the claims are patentable because a problem is encountered when rapid forward flight in helicopters is attempted and that the translational movement of the blades through the air interferes with the effect of the rotational movement of the blades, and that during the blades rearward sweep, they do not cut through the air, but rather travel at the same speed as the air passing by or have air pass over them in the same direction as the blade is moving. This argument is not persuasive because, contrary to Applicant's argument, the blades cut through the air during the blades rearward sweep because they are rotating, and hence cut through the air during both the blades forward sweep and the blade rearward sweep. With regard to Applicant's argument that a problem is encountered when rapid forward flight in helicopters is attempted and that the translational movement of the blades through the air interferes with the effect of the rotational movement of the blades, it appears that this problem would also affect Applicant's helicopter blades, because they are rotating in the same environment. Therefore, it is not seen how Applicant's blades would not be subject to this problem.

On page 2, lines 24-50 and page 3, lines 1-20, Applicant has provided two figures and has argued that the claims are patentable because in the present invention it is the shape swept out by the blades that has the properties of a lifting body, rather than the cross-section of the individual

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blade, as in the case of a standard helicopter. This argument is not persuasive. As explained during the interview of February 9, 2001, which is of record, and further reiterated below, the generation of lift in an airfoil is a result of the pressure differential between the upper and lower surfaces of the airfoil. Differences in curvature between the upper and lower surfaces cause such a pressure differential, which normally results in an upwardly directed force. The shape swept out by the blades during rotation clearly has the properties of a lifting body, for the reasons set forth above, and further because if the shape did not have the properties of a lifting body, as Applicant has argued, there would be no lift during blade rotation and forward flight, hence all helicopters using airfoil-shaped blades would drop out of the air because there would be no lift capable of sustaining the helicopter in air. Applicant's figure on page 2 shows a cambered airfoil, and all of the airfoils of Wallace, Wilford, Kingsbury, Hartt, Bennie, Black, and Kunz have at least some degree of camber. Because the airfoils are cambered, they will function according to the exact same principle of operation as Applicant's airfoil.

On page 3, lines 45-52 and page 4, lines 1-3, Applicant has argued that the PTO has not shown how any reference inherently shows a set of blades sweeping out the shape of a lifting body, and that the PTO should refer to and discuss the surface of revolution of the blade assembly, which the PTO has not done. The examiner respectfully strongly disagrees. The previous Office action clearly set forth all of the elements of the references which read on the claimed elements. In the event that Applicant did not understand the theory of inherency, the

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applied references all disclose rotatable blades 18 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. As mentioned above, because the airfoils are cambered, they will function according to the exact same principle of operation as Applicant's airfoil, thus generating lift. Note that anticipation by a prior art reference does not require either the inventive concept of the claimed subject matter or the recognition of inherent properties that may be possessed by the prior art reference. See, Verdegaal Bros. Inc. v. Union Oil Co., 814 F.2d 628, 633, 2 USPQ2d 1051, 1054 (Fed. Cir), cert. denied, 484 U.S. 827 (1987). A prior art reference anticipates the subject of a claim when the reference discloses every feature of the claimed invention, either explicitly or inherently (see, In re Paulsen, 30 F.3d 1475, 1478-1479, 31 USPQ2d 1671, 1675 (Fed. Cir. 1994), <u>In re Spada</u>, 911 F.2d 705, 708, 15 USPQ2d 1655, 1657 (Fed. Cir. 1990), Hazani v. Int'l Trade Comm'n, 126 F.3d 1473, 1477, 44 USPQ2d 1358, 1361 (Fed. Cir. 1997) and RCA Corp. v. Applied Digital Data Systems, Inc., 730 F.2d 1440, 1444, 221 USPQ 385, 388 (Fed. Cir. 1984). Moreover, a reference anticipates a claim if it discloses the claimed invention such that a skilled artisan could take its teachings in combination with his own knowledge of the particular art and be in possession of the invention. In re Graves, 69 F.3d 1147, 1152, 36 USPQ2d 1697, 1701 (Fed. Cir. 1995), cert. denied, 116 S. Ct. 1362 (1966), quoting from In re LeGrice, 301 F.2d 929, 936, 133 USPQ 365, 372 (CCPA 1962).

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Applicant has argued that in citing Wilford, it is unclear what shape that the blades sweep out and what about the shape gives it the characteristic of being a lifting body. The previous Office action clearly indicated that rotatable blades 10/11 sweep out the shape of a virtual disk. Note that any rotating object will sweep out the shape of a lifting body. With regard to Applicant's argument that the angled surface of Wilford is too sharply angled to be an effective lifting body, this argument is not persuasive because more of an angled surface results in more camber, which equates to greater lift. Because the blades are airfoil-shaped blades 18, lift will be generated as set forth above. With regard to Hartt and Wallace, Applicant has argued that a circular wing and not a virtual disk swept out by the blades is shown. This argument is not persuasive because rotation of the blades or circular wing about a vertical axis causes a virtual disk to be swept out. With regard to Applicant's argument that Kingsbury is directed to a propeller, rather than a set of helicopter blades, note that column 1, lines 9-10 states that the invention relates to a propeller for aircraft. The preamble of claim 1 recites a helicopter blade assembly, which is a quasi-intended use recitation. The propeller of Kingsbury is clearly usable as a propeller for a helicopter. A recitation of the intended use of the claimed invention must result in a structural difference between the claimed invention and the prior art in order to patentably distinguish the claimed invention from the prior art. If the prior art structure is capable of performing the intended use, then it meets the claim. In a claim drawn to a process of making, the intended use must result in a manipulative difference as compared to the prior art. See In re Casey, 152 USPQ 235 (CCPA 1967) and In re Otto, 136 USPQ 458, 459 (CCPA 1963).

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With regard to Applicant's argument that Bennie shows a standard helicopter blade set except that there is a mechanism for altering the pitch of the outer portion of the blade, which does not affect the camber and is not a lifting body, these arguments are not persuasive because as set forth in the previous Office action, means 58 control the aerodynamic warping of the blades, which is the same as the camber. Blades 41, 42 are clearly airfoil-shaped, and thus causing lift during rotation of the virtual body. With regard to Applicant's argument that lever 15 of Wilford changes the hub and axis of rotation which is different from camber, lever 15 is moved to tilt the hub and axis of rotation, which will shift the position of outer portion 11 of the blades due to the change in the forces of lift. See page 2, lines 27-31 and 69-75.

Drawings

The drawings are objected to because figures 1-12 are replete with elements that are shown in cross section which must be indicated by hatching. Correction or proposed correction is required and examiners may no longer permit drawing corrections to be held in abeyance. See 37 C.F.R. 1.121(d).

Specification

The abstract of the disclosure is objected to because it contains the term "is disclosed" (line 7) which is implied and should be deleted. Correction is required. See MPEP § 608.01(b).

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Claim Rejections - 35 USC § 112

The following is a quotation of the second paragraph of 35 U.S.C. 112:

The specification shall conclude with one or more claims particularly pointing out and distinctly claiming the subject matter which the applicant regards as his invention.

Claim 4 is rejected under 35 U.S.C. 112, second paragraph, as being indefinite for failing to particularly point out and distinctly claim the subject matter which applicant regards as the invention. In claim 4, lines 3 and 4, the terms "substantially" and "gently" are indefinite. It is established patent law that the claims must define the metes and bounds of the invention with a reasonable degree of certainty. In re Venezia, 530 F.2d 956, 958, 189 USPQ 149, 151 (CCPA 1976). When a word of degree such as "substantially" or "gently" is used in a claim, it must be determined whether the underlying specification provides some standard or guideline for measuring that degree, such that a person of ordinary skill in the art would understand what is claimed when the claim is read in light of the specification. Seattle Box Co. V. Industrial Crating & Packing Inc., 731 F.2d 818, 826, 221 USPQ 568, 574 (Fed. Cir. 1984). A review of the specification finds no standards or guidelines in Applicants' specification for measuring the scope of the words "substantially" and "gently". Hence, the terms "substantially" and "gently" are indefinite and the metes and bounds of this term cannot be determined with a reasonable degree of certainty.

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Claim Rejections - 35 USC § 102

The following is a quotation of the appropriate paragraphs of 35 U.S.C. 102 that form the basis for the rejections under this section made in this Office action:

A person shall be entitled to a patent unless --

(b) the invention was patented or described in a printed publication in this or a foreign country or in public use or on sale in this country, more than one year prior to the date of application for patent in the United States.

Claims 1 and 4, as far as claim 4 is definite, are rejected under 35 U.S.C. 102(b) as being anticipated by Wallace (figure 2). Note the helicopter blade assembly 17 permitting rapid forward flight with separate means 13 for providing a forward impetus, the substantially vertical mast 35, and rotatable blades 18 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. The virtual disk swept out has a center 19 (figure 4) which is substantially flat at and near the center, and slopes downwardly at the edge of the virtual disk near 18.

Claims 1-4, as far as claim 4 is definite, are rejected under 35 U.S.C. 102(b) as being anticipated by Wilford (figure 5). Note the helicopter blade assembly 5 permitting rapid forward flight with separate means 3 for providing a forward impetus, the substantially vertical mast (unnumbered, near 4), and rotatable blades 10/11 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. Note the means for controlling the camber

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15. The blade outward tips (near 11) have their camber controlled by downward bending near the outward tips of the blades. The virtual disk swept out has a center 19 (figure 4) which is substantially flat at and near the center, and slopes downwardly at the edge of the virtual disk near 11.

Claims 1-3 are rejected under 35 U.S.C. 102(b) as being anticipated by Kingsbury (figures 2-3). Note the helicopter blade assembly (unnumbered) permitting rapid forward flight, the substantially vertical mast 1, and rotatable blades 16, 17 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. Note the means for controlling the camber 5/20. The blade outward tips (near 16, 17) have their camber controlled by downward bending near the outward tips of the blades.

Claim 1 is also rejected under 35 U.S.C. 102(b) as being anticipated by Hartt (figure 2).

Note the helicopter blade assembly 46 permitting rapid forward flight with separate means 36 for providing a forward impetus, the substantially vertical mast 26, and rotatable blades 46 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift.

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Claims 1-3 are also rejected under 35 U.S.C. 102(b) as being anticipated by Bennie. Note the helicopter blade assembly near 41, 42 permitting rapid forward flight, the mast 60 which is considered to be "substantially vertical" because it extends mainly in a vertical direction, and rotatable blades 41, 42 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. Note the means 58 for controlling the aerodynamic warping of the blades, which is equivalent to controlling the camber. The blade outward tips have their camber controlled by downward bending near the outward tips of the blades.

Claim 1 is also rejected under 35 U.S.C. 102(b) as being anticipated by Black (figures 1-2 and 4). Note the helicopter blade assembly 12 permitting rapid forward flight with separate means 22 for providing a forward impetus, the substantially vertical mast near 86, and rotatable blades 40, 44 which sweep out the shape of a virtual disk having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift.

Claims 1 and 4, as far as claim 4 are definite, are also rejected under 35 U.S.C. 102(b) as being anticipated by Kunz (figures 1 and 3). Note the helicopter blade assembly 10 permitting rapid forward flight with separate means 62 for providing a forward impetus, the substantially vertical mast near 32, and rotatable blades 38/42/26' which sweep out the shape of a virtual disk

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having the properties of a lifting body when rapidly rotated by the mast so that as the disk is pushed translationally through the air, it generates lift. The virtual disk swept out has a center near 32 (figure 3) which is substantially flat at and near the center, and slopes downwardly at the edge of the virtual disk near 26'.

Applicant's amendment necessitated the new ground(s) of rejection presented in this Office action. Accordingly, **THIS ACTION IS MADE FINAL**. See MPEP § 706.07(a). Applicant is reminded of the extension of time policy as set forth in 37 CFR 1.136(a).

A shortened statutory period for reply to this final action is set to expire THREE MONTHS from the mailing date of this action. In the event a first reply is filed within TWO MONTHS of the mailing date of this final action and the advisory action is not mailed until after the end of the THREE-MONTH shortened statutory period, then the shortened statutory period will expire on the date the advisory action is mailed, and any extension fee pursuant to 37 CFR 1.136(a) will be calculated from the mailing date of the advisory action. In no event, however, will the statutory period for reply expire later than SIX MONTHS from the date of this final action.

Any inquiry concerning this communication or earlier communications from the examiner should be directed to Examiner Verdier whose telephone number is (703) 308-2638. The examiner can normally be reached on Monday-Friday from 9:00 a.m. to 5:30 p.m.

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If attempts to reach the examiner by telephone are unsuccessful, the examiner's supervisor, Edward Look, can be reached on (703) 308-1044. The fax phone number for this Group is (703) 305-3588.

Any inquiry of a general nature or relating to the status of this application or proceeding should be directed to the Group receptionist whose telephone number is (703) 308-0861.

CV June 1, 2001 Christopher Verdier Primary Examiner Art Unit 3745